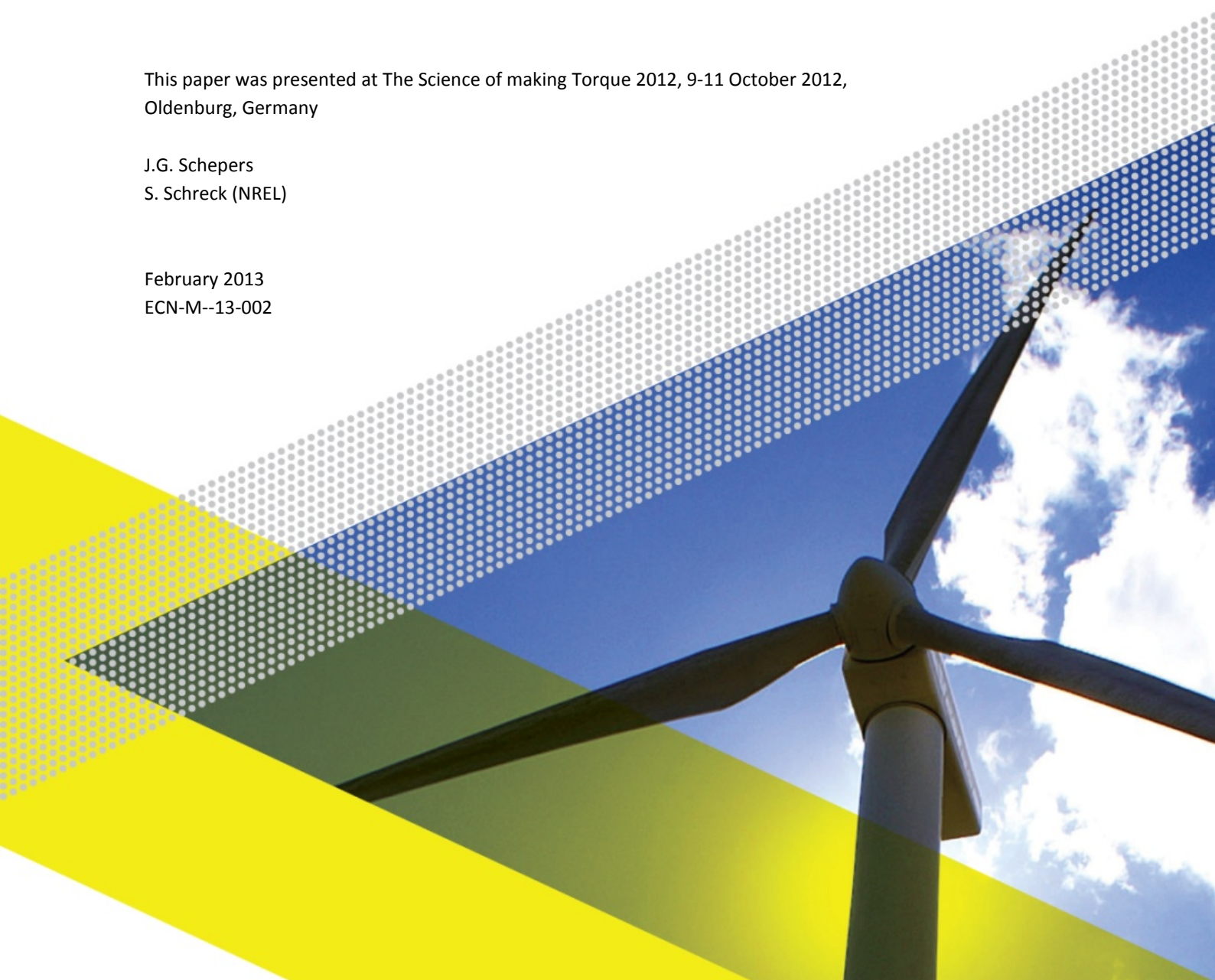


The importance of aerodynamics and the role of aerodynamic measurements

This paper was presented at The Science of making Torque 2012, 9-11 October 2012,
Oldenburg, Germany

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ECN-M--13-002



The importance of aerodynamics and the role of aerodynamic measurements

Illustrated with results from the NASA-Ames experiment at yaw

Gerard Schepers (ECN) and Scott Schreck (NREL)

The science of Making Torque, Oldenburg

Content

- *Importance of wind turbine aerodynamics*
 - Aerodynamic design modelling
 - Aerodynamic measurements
- Aerodynamic wind turbine measurements
 - IEA projects on field and wind tunnel measurements (IEA Task 14, IEA Task 18, IEA Task 20 and IEA Task 29 (Mexnext-I))
 - Power at yawed conditions in NASA-Ames measurements
 - Current and future developments with regard to aerodynamic measurements

AERODYNAMICS OF WIND TURBINES



Important for:

- Energy production
- Loads (strength, costs!)
- Stability (failure, damage)
- Control
 - Stall
 - (Individual) pitch
 - Distributed aerodynamic control

And hence the overall success of a wind turbine design

AERODYNAMICS

- Aerodynamics of wind turbines is extremely difficult
 - Rotating
 - In lower part of atmosphere → extremely turbulent
 - Instationary
 - Stall(!)
 - Large variety in scales (Diameter can be twice span of Airbus A380!)
 - Constraints and interactions (e.g. costs, system dynamics)
 - Wind turbine aerodynamic calculations are extremely time consuming
 - Load calculations: Time domain, many long time series needed to get ‘statistics right’
 - ($\sim 10^6$ nr of time steps) → Computational time of a design spectrum can easily be longer than lifetime of the wind turbine
 - Generally calculations are done with simplified **Blade Element Momentum (BEM)** Theory with engineering add-ons to cover instationary effects, stall, 3D effects, yaw etc.

Aerodynamic modelling

- Former validation projects generally show 10-20% accuracy of blade loads calculated with **BEM** for **standard** conditions, (excluding stall) but still large uncertainties (i.e. 50%) in off-design conditions ¹⁾
- What about uncertainties for modern wind turbines?
 - Wind turbine designs become ‘**unconventional**’, what about the modelling aspects?
 - Aerodynamic modelling of distributed control **devices**
 - For very **large** turbines BEM may not model correctly:
 - Incoherent structures over the rotor
 - Extreme shear, low level jets, 3D shear effects (wind farms!)
- Research needed to improve the accuracy, among others by other types of modelling, e.g **CFD**

1) Schepers, J.G.; Heijdra, J.J.; Thomsen, K.; Larsen, T.; Foussekis, D.; Rawlinson Smith, R.; Kraan, I.; Visser, B.; Øye, S.; Ganander, H.; Carlen, I.; Voutsinas, S.; Belessis, M.; Drost, L. Verification of European Wind turbine design codes, presented at European wind energy conference and exhibition, Copenhagen, Denmark, 2-6 july, 2001.

Aerodynamics

- **MEASUREMENTS , MEASUREMENTS, MEASUREMENTS**
- Detailed aerodynamic measurements along the blades are urgently needed at several conditions

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IEA TASKS ON AERODYNAMIC MEASUREMENTS



- 1991-1997: IEA Task 14 (Field Rotor Aerodynamics, Operating Agent: ECN)
- 1997-2001: IEA Task 18 (Field Rotor Aerodynamics, enhanced, Operating Agent: ECN))
- 2001-2007: IEA Task 20: (Analysis of NREL's NASA-Ames, measurements, Operating Agent: NREL)
- 2008-2011: IEA Task 29: Mexnext-I (Analysis of Mexico measurements, Operating Agent: ECN)
- 2011-2014: IEA Task 29: Mexnext-II (various measurements, Operating Agent, ECN)

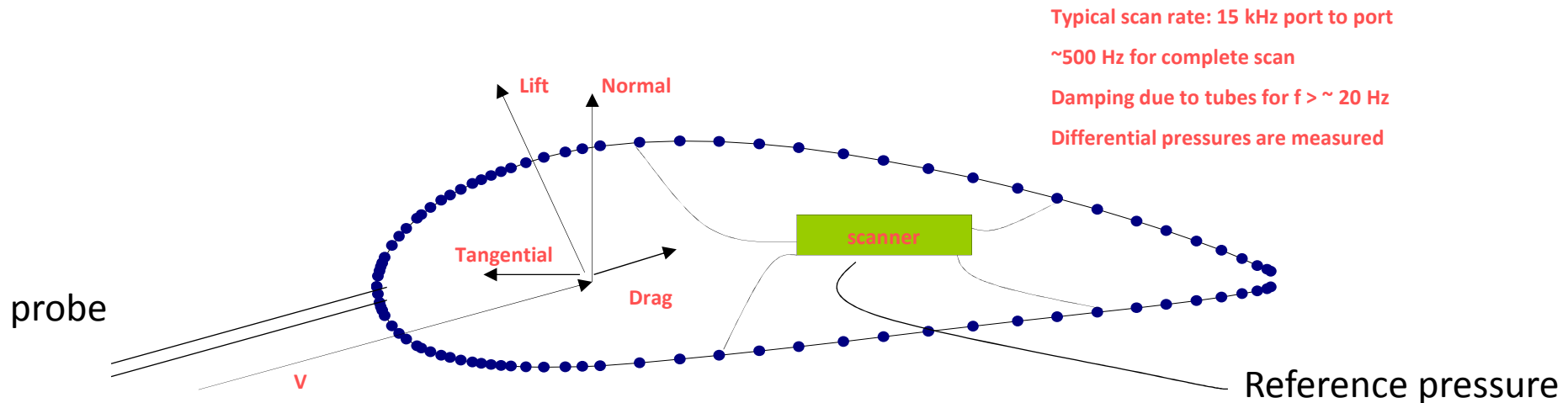
Rotor aerodynamic measurements: IEA Tasks 14/18



- Period: 1991-2001
- Aim: To coordinate aerodynamic test programs on field facilities
- Participants
 - Netherlands Energy Research Foundation, ECN (NL; Operating Agent)
 - Delft University of Technology, DUT (NL)
 - Imperial College/Rutherford Appleton Laboratory, IC/RAL (UK, Only Task 14)
 - National Renewable Energy Laboratory, NREL (USA)
 - RISØ, The Test Station for Wind Turbines (DK)
 - Mie University (JP, Only Task 18)
 - Centre for Renewable Energy Sources, CRES (Gr, Only Task 18)

Aerodynamic measurements

- To develop, validate aerodynamic models
- Conventional measurement programs: Only indirect, global aerodynamic information
- Desired: Direct local aerodynamic properties (I.e. pressure distributions, inflow angles, inflow velocities)



IEA Tasks 14/18: Facilities



ECN facility

3-2-2013



NREL facility

IEA Task 14/18 some results

- Database on <http://www.ecn.nl/nl/units/wind/projecten/field-rotor-aerodynamics-database/>
- ‘Discovery’ of
 - Stall delay on wind turbines (underprediction of loads at inner part of the blade at large angles of attack when using 2D airfoil coefficients)
 - Overprediction of tip loads when using 2D airfoil coefficients
 - ‘Compensating’ errors when using global measurements

Validation measurements, status at end of 90's

1990:

Measurements of **power** and **loads** showed differences but they were too global to form a basis for improvement of aerodynamic models

Desired:

- Local aerodynamic loads (pressure distribution) in field conditions (**IEA Tasks 14/18**)
- But also: Constant, uniform and controlled conditions (**→Windtunnel**)

Measurements in NASA-Ames wind tunnel



- Carried out by NREL (National Renewable Energy Laboratory), USA
- Spring 2000
- 24m x 36m NASA-Ames wind tunnel.
- 10 m rotor
- Measurement of pressure distributions at 5 locations along rotor blade
- Analysed in IEA Task 20 (Operating Agent: NREL)
- Participants:
 - ETS (Canada)
 - RISO/DTU (Denmark)
 - CRES/NTUA (Greece)
 - ECN/TU Delft (Holland)
 - IFE (Norway)
 - CENER (Spain)
 - HGO (Sweden)



3-2-2013



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 - Current and future developments

Illustration of the value of aerodynamic measurements



A study of the power at yawed conditions in NASA-Ames

Most of the attention (also in the IEA tasks) was focussed on **loads** at yawed conditions

- What happens with the **power** in yaw?
- It is proposed to assume a cosinoidal behaviour (inspired by ¹⁾)
 - $P \sim P_0 \cos^x \phi_y$ with P_0 the power at zero yaw and x the exponent to be determined

1) Dahlberg, J.A. and Montgomerie, B

Research program of the Utgrunden Demonstration Offshore Wind Farm, Final report Part 2, Wake effects and other loads

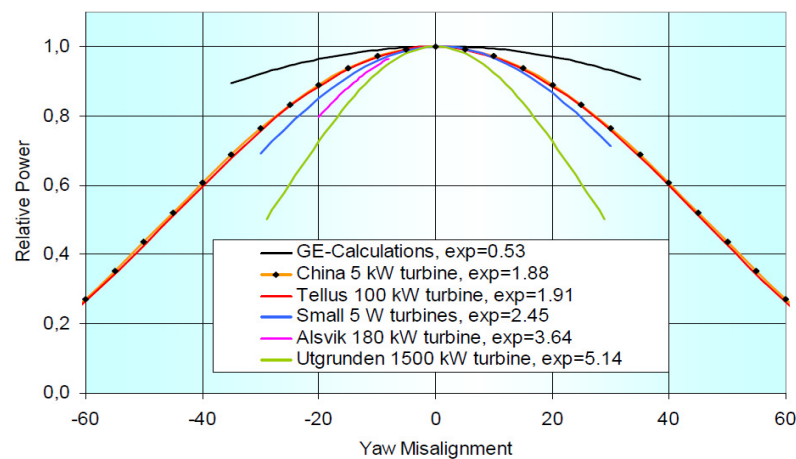
Swedish Defense Research Agency, FOI, FOI 2005-02-17 , 2005

$x = 3?$

- Some people think that $x = 3$: $P \sim P_0 \cos^3 \phi_y$
based on the assumption that:
 - $P \sim V_{\text{axial}}^3$
 - $V_{\text{axial}} = V_w \cos(\phi_y)$
 - This (implicitly) assumes a constant *induction* with yaw angle
- However *induction* changes with yaw angle!

Research from Jan-Ake Dahlberg ¹⁾

- Jan Ake Dahlberg (Vattenfall/FFA) finds on basis of field and wind tunnel measurements that:
 - $P \sim P_0 \cos^x \phi_y$ with x to vary between **1.88** and **5.15**
 - BEM **overpredicts** the power at yawed conditions

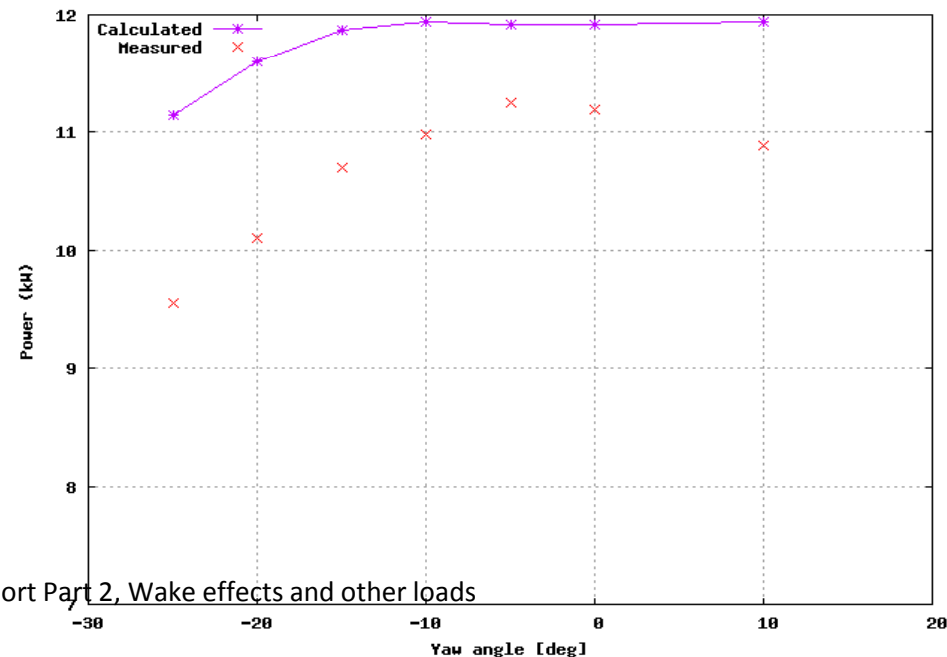


¹⁾ Dahlberg, J.A. and Montgomerie, B
 Research program of the Utgrunden Demonstration Offshore Wind Farm,
 Final report Part 2, Wake effects and other loads
 Swedish Defense Research Agency, FOI, FOI 2005-02-17 , 2005

DATA measurements on rotor with $D=4.5$ m in German Dutch Wind Tunnel DNW at design conditions ¹⁾



- $P \sim P_0 \cos^x \phi_y$ with $x \sim 1.8$
- BEM **overpredicts** the power at yawed conditions
- Generally consistent with findings from ²⁾



¹⁾ J.G. Schepers EU project in German Dutch Wind Tunnel ECN-RX--01-006, 2001

²⁾ Dahlberg, J.A. and Montgomerie, B

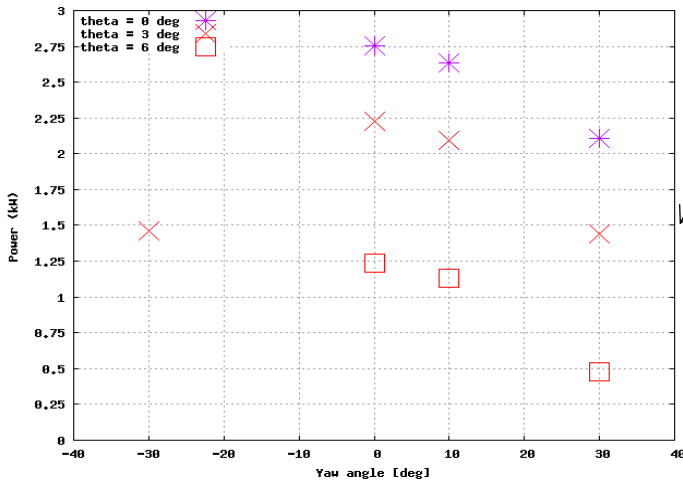
Research program of the Utgrunden Demonstration Offshore Wind Farm, Final report Part 2, Wake effects and other loads

Swedish Defense Research Agency, FOI, FOI 2005-02-17 , 2005

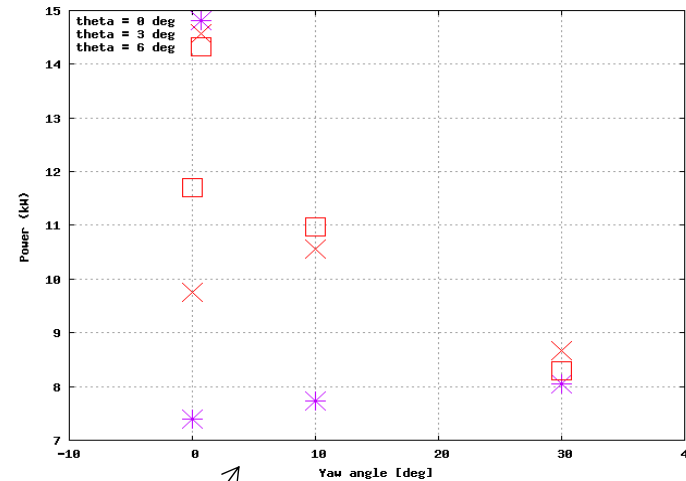
Analysis on NREL Phase VI(NASA-Ames) $P(\phi_y)$ measurements



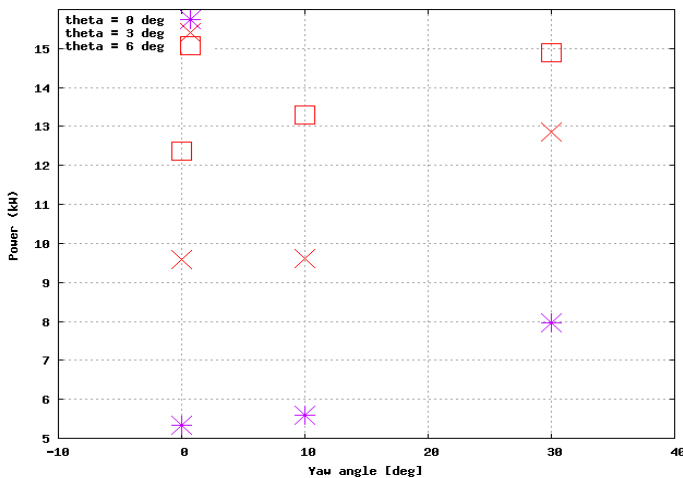
Results analyzed at $V_{tun} = 5, 10$ and 15 m/s and $\theta = 0, 3$ and 6 degrees



$V_{tun} = 5$ m/s: Power decreases with yaw



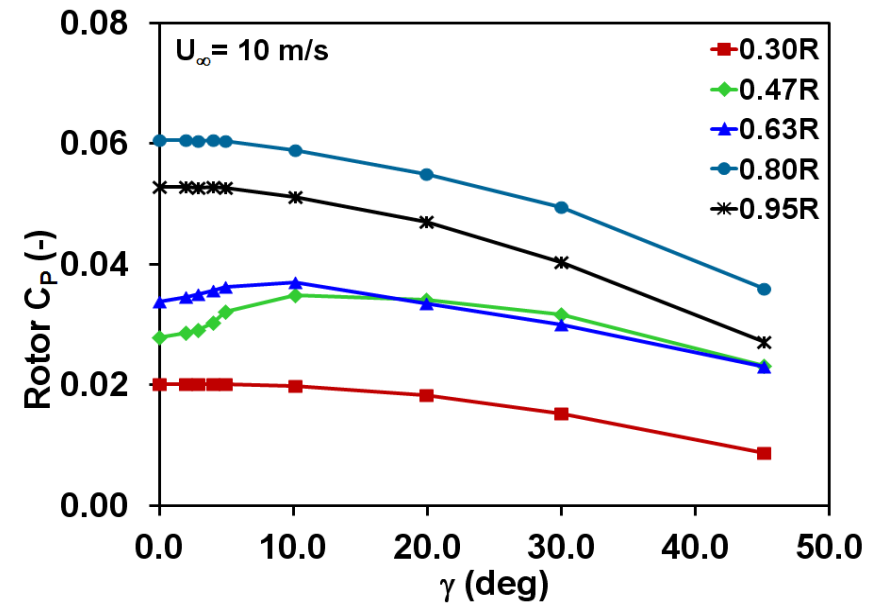
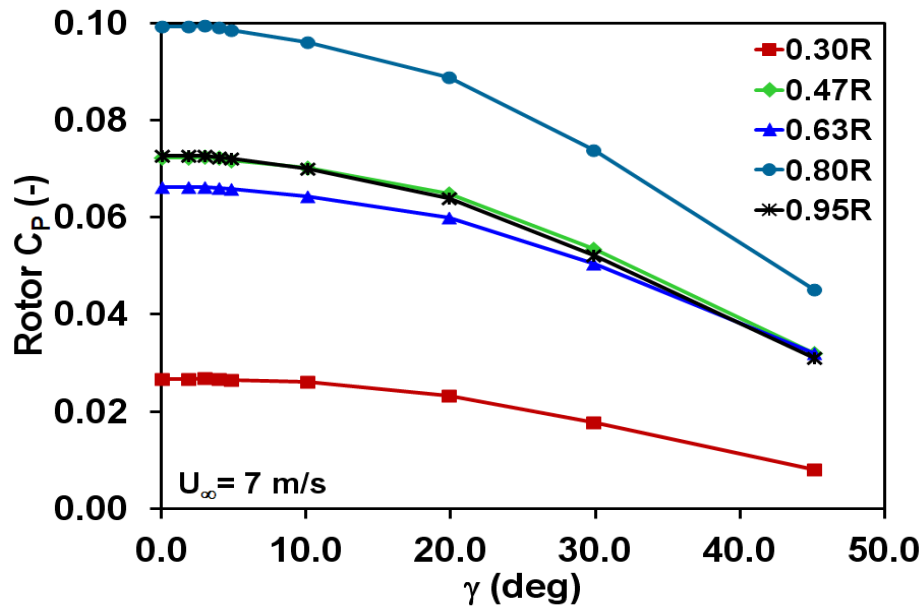
$V_{tun} = 10$ m/s: Power decreases with yaw for $\theta = 6$ deg
increases with yaw for $\theta = 0$ deg



$V_{tun} = 15$ m/s: Power increases with yaw

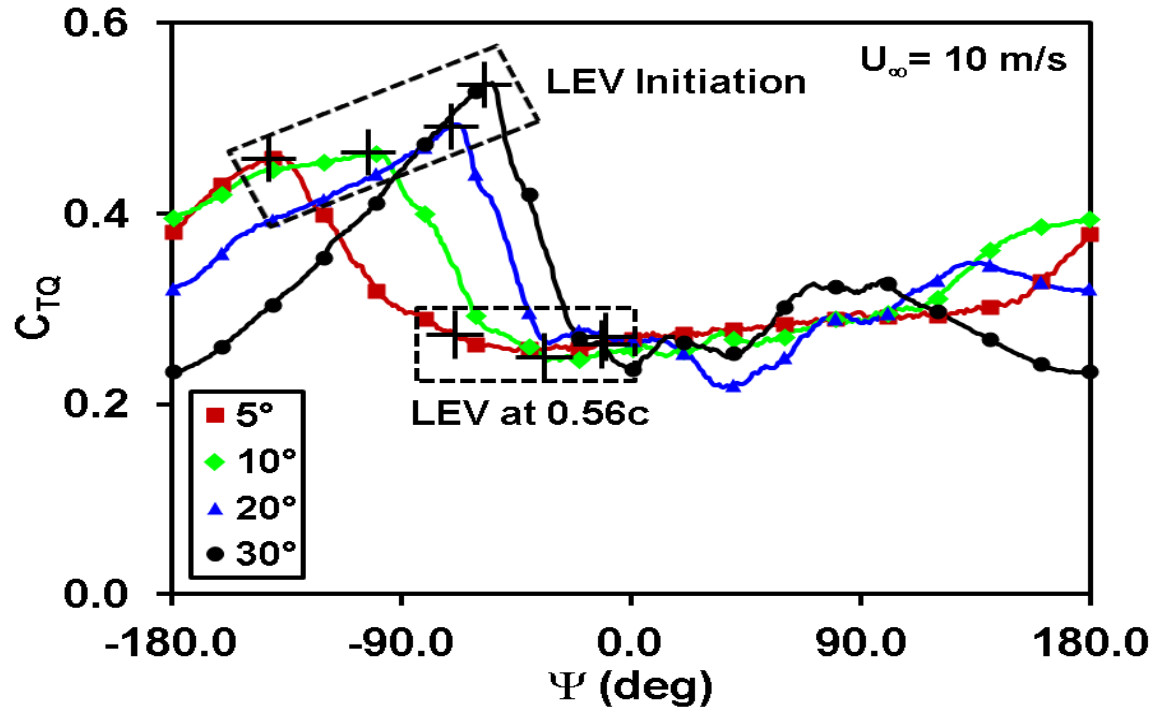
- Trend: Power decrease with yaw seems less apparent for **larger angles of attack** (high wind speed and low pitch angle) and eventually the power even **increases** with yaw!

Analysis on NREL Phase VI(NASA-Ames) $C_p(\phi_y)$ measurements



- Sectional C_p as function of yaw angle at 7 m/s and 10 m/s
- Conclusion:
 - 7 m/s: Monotonic decrease
 - 10 m/s: Initial increase at 47% and 63% span

Analysis on NREL Phase VI(NASA-Ames) measurements at yaw



Sectional torque coefficient as function of azimuth angle at different yaw angles

Conclusion:

- Strong unsteady aerodynamic effects dependant on yaw angle
- Non-linear unsteady aerodynamic effects determine the rotor averaged torque (i.e. power)

Value of aerodynamic wind tunnel measurements because:

- The $P(\phi_y)$ behaviour could be determined at well defined conditions
- The sectional $P(\phi_y)$ behaviour could be determined
- The unsteady airfoil aerodynamics could be determined at yaw
- The conclusion can be drawn that high wind speeds, low pitch angles, and the inner part of the blade makes a power increase with yaw angle more likely
→ large angles of attack lead to a power increase due to unsteady airfoil aerodynamics
- Remaining question: Why does BEM overpredict the power at yawed conditions
 - Possibly related to the generally accepted calculation of disc averaged induction according to Glauert:

$$F_{ax} = \rho A_r \left| \vec{V}_w + \vec{u}_{i0} \right| \cdot 2 \vec{u}_{i0}$$

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Validation measurements, status at ~2005

1990:

Measurements of **power** and **loads** showed differences but they were too global to form a basis for improvement of aerodynamic models

Desired:

- Local aerodynamic loads (pressure distribution) in field conditions (**IEA Tasks 14/18**)
- Constant, uniform and controlled conditions (→ **NASA-Ames windtunnel measurements from IEA Task 20**)
- But also: Induced velocities and wake velocities (→ Detailed flow field measurements from **Mexico** project)

Rotor aerodynamic measurements: EU project Mexico

Model rotor EXperiments In Controlled conditions ¹⁾



- Coordinated by ECN
- 2001-2006
- Measurements in German Dutch Wind tunnel, DNW



- North East Polder (Netherlands)
- Open test section: $9.5 \times 9.5 \text{ m}^2$
- Diameter of rotor: 4.5 m
 - Fast pressure measurements at 5 positions (25%, 35%, 60%, 82% and 92% span) along the blade
 - Particle Image Velocimetry (PIV): Quantitative flow visualisation



¹⁾ Participants: see <http://www.ecn.nl/nl/units/wind/rd-programma/aerodynamica/projects/mexico/>

Rotor aerodynamic measurements: IEA Task 29 MexNext-I: Goals and Participants

- Goal: Analysis of Mexico measurements
- Participation from the following institutes from 11 different countries:
 - Canada (École de technologie supérieur, Montreal (ETS), University of Victoria (UVic))
 - Denmark(DTU-RISO/DTU(Mek))
 - Germany(University of Stuttgart (IAG), University of Applied Sciences, Kiel, Forwind)
 - Israel (Israel Institute of Technology (Technion))
 - Japan (Mie University/National Institute of Advanced Industrial Science)
 - Korea((Korea Institute of Energy Research (Kier) and Korea Aerospace Research Institute (Kari))
 - Netherlands(Energy Research Center of the Netherlands (ECN), University of Delft (TUDelft), Technical University of Twente, Suzlon Blade Technology)
 - Norway (Institute for Energy Technology/Norwegian University of Science and Technology (IFE/NTNU))
 - Spain(Renewable Energy National Centre of Spain (CENER) and National Institute for Aerospace Technology, INTA)
 - Sweden(Royal Institute of Technology/University of Gotland (KTH/HGO))
 - USA (National Renewable Energy Laboratory (NREL))

Results from Mexnext-I

- Has already been presented where it is mainly the combination of flow field measurements and detailed aerodynamic measurements on the blade which made the Mexico experiment unique

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Current and planned developments on the ECN field of aerodynamic measurements

- IEA Task 29, Mexnext-II:
 - Unexplored aerodynamic data from several facilities (both wind tunnel and field) to be discussed at
 - New Mexico measurements in DNW wind tunnel ¹⁾ based on the existing measurement set-up to complement the database
- UAE Phase VII: New measurements in NASA-Ames wind tunnel
- TUDelft wind tunnel measurements
- Mie wind tunnel measurements ----- and some more wind tunnel measurements
- Dan-Aero: Danish field aerodynamic measurements
- EERA (European Energy Research Alliance) : New field facility

¹⁾Acknowledgment to ESWIRP and Inwind

DU Windtunnels for aerodynamic research **ECN** on wind turbines (Courtesy Gerard van Bussel)



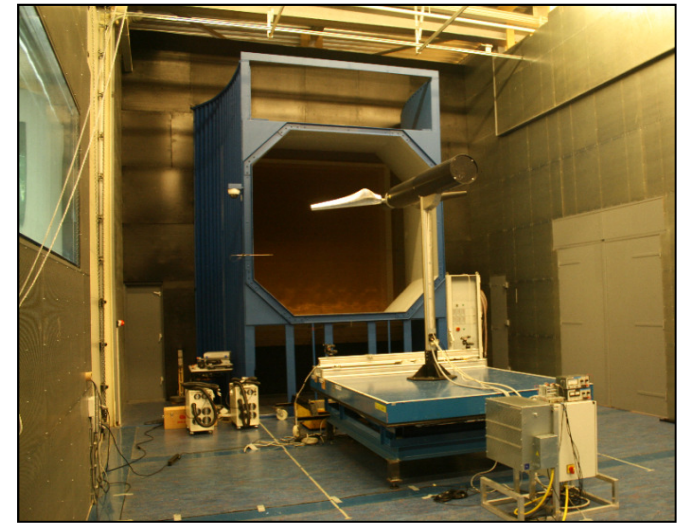
LSTL tunnel
 $V_{\max} = 110$ m/s

test section 1.2 x 1.85 m
mainly for airfoil development



Open Jet tunnel
 $V_{\max} = 15$ m/s

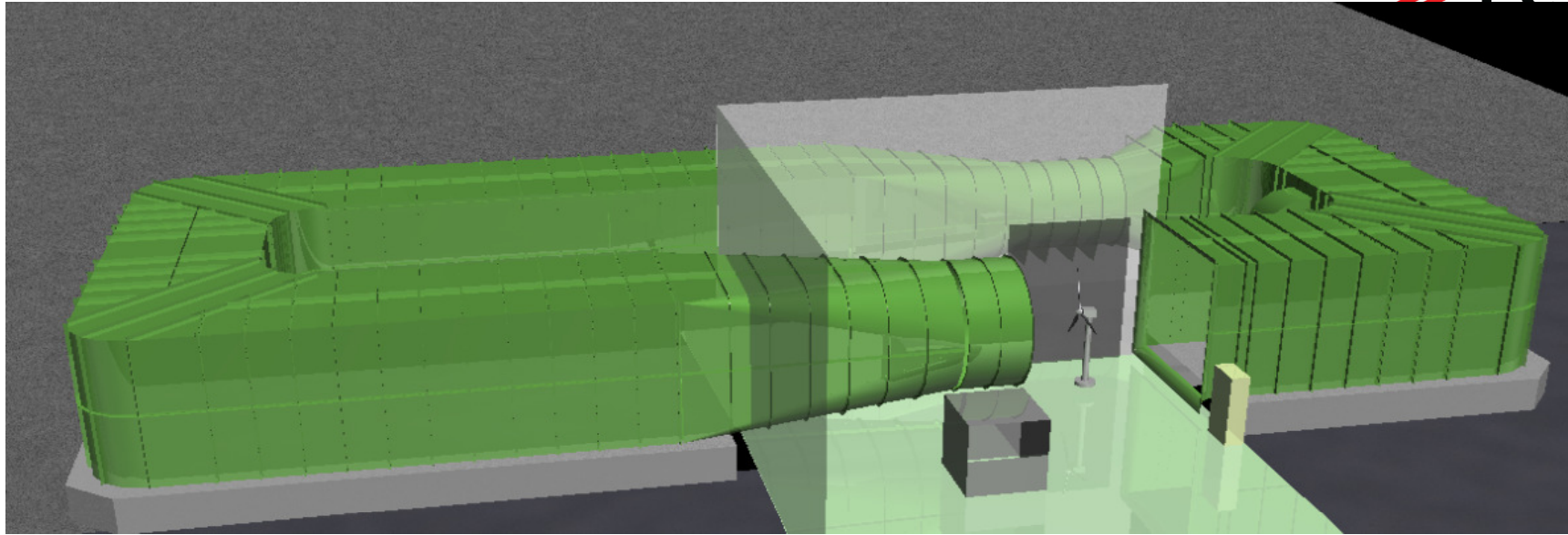
test section 2.2 m O/
mainly for rotor flow analysis



Open Jet Facility (2009)
 $V_{\max} = 35$ m/s

test section 3.0 m O/ 13 m long
rotor- and unsteady flow analysis

Mie University Wind Tunnel



Single return wind tunnel with open test section

Outlet diameter 3.6m

or 3.0m using high contraction nozzle

Suction section 4.5m x 4.5m rectangular cross-section

Length of test section 4.5m

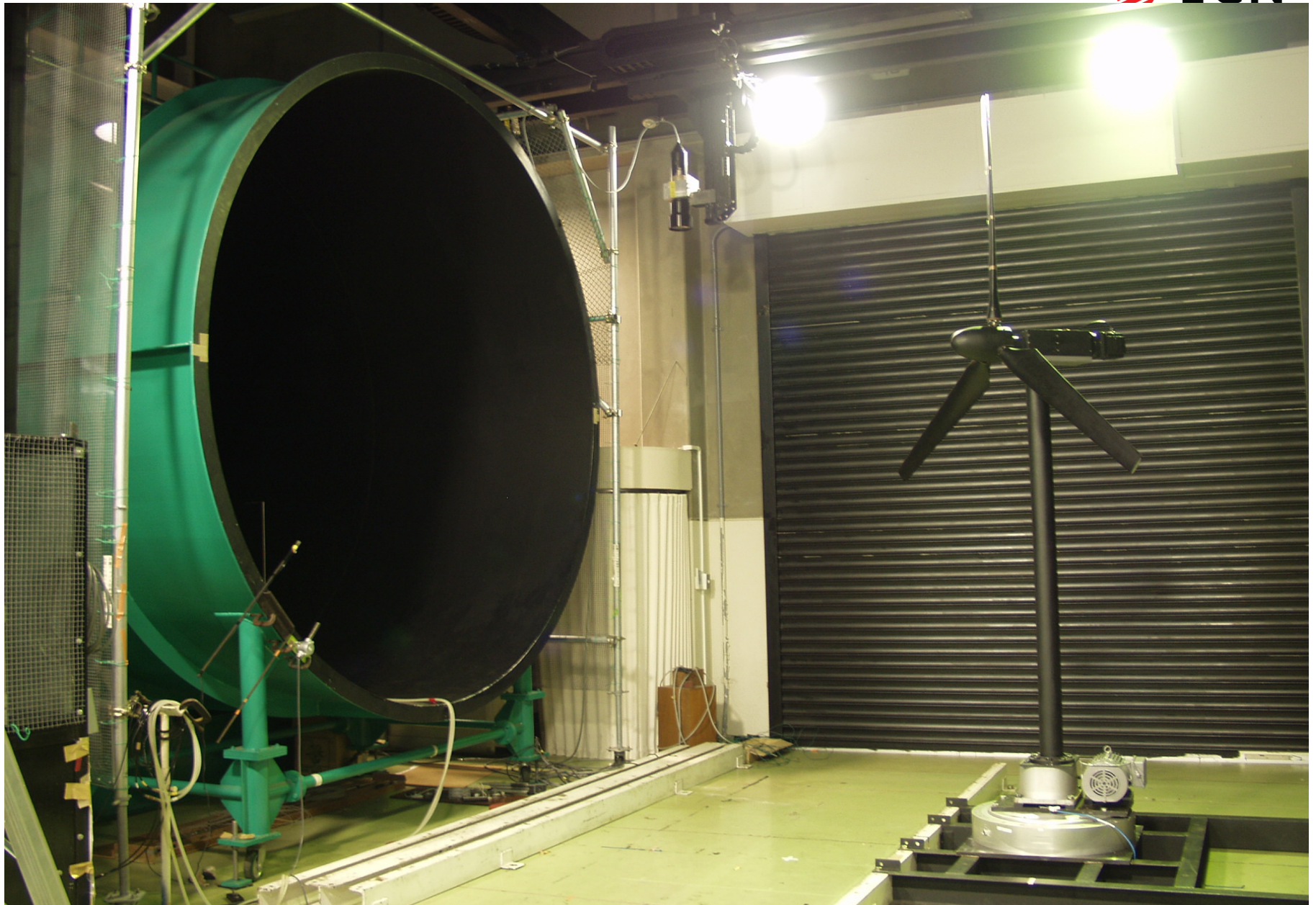
Wind speed 0 - 30m/s for 3.6m nozzle

Uniformity < 1.5% (7m/s)

Turbulence level < 0.5% (7m/s)

Measurements e.g. pressure measurements along blade, LDA

Rotor with diameter of 3.4 meter



EERA Field experiment, background

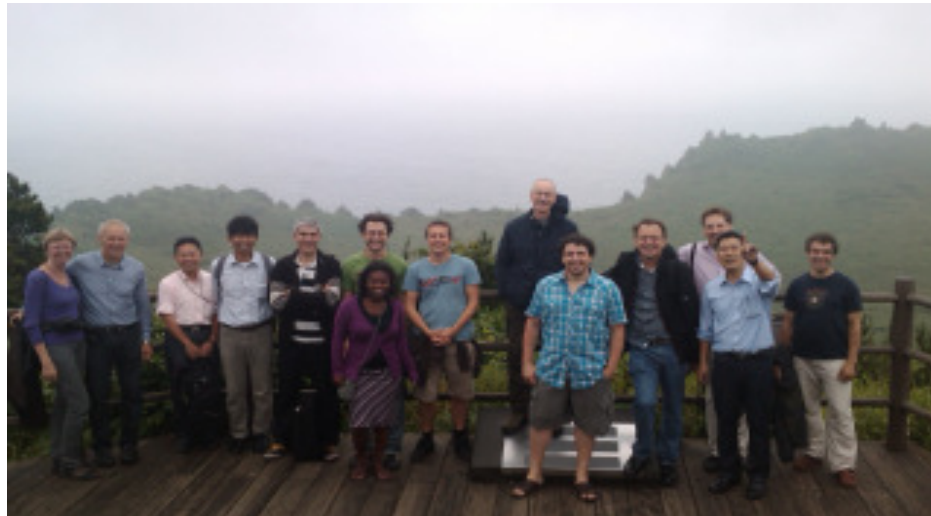


- **Wind tunnel** measurements and **field** measurements should be seen as **COMPLEMENTARY**
 - Wind tunnel measurements are done at well known (but less representative) conditions
 - This facilitates the interpretation of data and enables the determination of e.g. the $P(\phi_y)$ behaviour
 - Field measurements are done at representative (but less known) conditions
- We see initiatives on wind tunnel measurements (e.g. *New Mexico/NASA-Ames/TU Delft*) but not on field measurements:
- **The time is ripe for a PUBLIC aerodynamic field experiment on a wind turbine representative for the current and future generation of wind turbines based on:**
 - Most innovative measurement techniques
 - Carried out under auspices of EERA (European Energy Research Alliance)

Conclusions

- Since the end of the 80's/beginning of the 90's dedicated aerodynamic measurements have been performed on wind turbines both in the field and the wind tunnel.
- The level of maturity of these measurements has been improved enormously over the years.
- The measurements have delivered a huge amount of data and knowledge by which design codes could be improved and validated which led to more reliable and efficient wind turbine designs.

Questions?



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